

# 1 Rotating structures and Bryan's effect

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6 In 1890 Bryan observed that when a vibrating structure is rotated the vibrating pattern rotates at a  
7 rate proportional to the rate of rotation. During investigations of the effect in various solid and  
8 fluid-filled objects of various shapes, an interesting commonality was found in connection with the  
9 gyroscopic effects of the rotating object. The effect has also been discussed in connection with a  
10 rotating fluid-filled wineglass. A linear theory is developed, assuming that the rotation rate is  
11 constant and much smaller than the lowest eigenfrequency of the vibrating system. The associated  
12 physics and mathematics are easy enough for undergraduate students to understand. © 2009 American  
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## 15 I. INTRODUCTION

16 When a vibrating structure is subjected to a rotation at an  
17 angular rate  $\Omega$ , the vibrating pattern rotates with respect to  
18 the structure at a rate proportional to  $\Omega$ . This effect, known  
19 as “Bryan's effect,” was first observed by Bryan<sup>1</sup> in 1890.  
20 Bryan defined the constant of proportionality for a body con-  
21 sisting of a ring or cylinder for various modes of vibration as

$$22 \text{BF} = \frac{\text{Angular rate of the vibrating pattern}}{\text{Angular rate of the vibrating body}}. \quad (1)$$

23 The constant of proportionality BF is known as “Bryan's  
24 factor.” Estimates based on Bryan's effect were used to dem-  
25 onstrate that the resonance of liquid surface vibrations in a  
26 wineglass<sup>2</sup> was predictable using a membrane model.<sup>3</sup>

27 We have been investigating Bryan's effect in various solid  
28 and fluid-filled symmetric objects that rotate at a constant  
29 rate which is much smaller than the lowest frequency of  
30 vibration of the structure. To understand Bryan's effect, in-  
31 vestigations were conducted starting with a slowly rotating  
32 and vibrating (isotropic solid) disc and then progressing to a  
33 cylinder and a sphere. Each of these investigations yielded  
34 identical (up to constant coefficients) ordinary differential  
35 equations that can be used to explain Bryan's effect. In this  
36 paper we demonstrate how these differential equations are  
37 derived, how Bryan's factor can be calculated, and how  
38 Bryan's effect can be predicted.

39 In 1988, Zhuravlev and Klimov<sup>4</sup> investigated Bryan's ef-  
40 fect for an isotropic, spherically symmetric body rotating in  
41 three dimensions. Among other results, they demonstrated  
42 that Bryan's effect depends on the vibration mode. Bryan's fac-  
43 tor has numerous navigational applications.<sup>5</sup> Bryan's fac-  
44 tor is used to calibrate vibrating cylindrical gyroscopes. In  
45 Ref. 5 a thin cylindrical shell was considered for both high  
46 and low rotational rates. Apart from navigational applica-  
47 tions, the theory presented in Ref. 6 could be useful in un-  
48 derstanding the dynamics of pulsating stars and earthquakes.

49 We will discuss Bryan's effect for a symmetrically distrib-  
50 uted annular disc, where both radial and tangential vibrations  
51 are considered, and ignore axial vibrations. The theory is  
52 readily adapted to an isotropic solid cylinder (or sphere) in  
53 the form of concentric cylindrical (or spherical) bodies where  
54 some of the layers are fluids.

## II. TRUE VELOCITY

55

56 Consider a body consisting of a solid disk with distributed  
57 parameters as depicted in Fig. 1. Let  $N$  be the number of  
58 concentric annular layers in the system and  $a_{i-1}$  and  $a_i$  the  
59 inner and outer radii of the  $i$ th annulus each with density  $\rho_i$ ,  
60 thickness  $h_i$ , modulus of elasticity  $E_i$ , and Poisson's ratio  $\nu_i$ ,  
61  $i = 1, \dots, N$  [see Eqs. (A3) and (A4)]. Assume that the disk is  
62 subjected to nondecaying tangential and radial vibrations in  
63 one of its natural modes and that vibration is absent along the  
64  $z$ -axis. In polar coordinates (with  $x = r \cos \varphi$  and  $y = r \sin \varphi$ )  
65 consider the equilibrium position  $(x, y) \equiv P(r, \varphi)$  of a vibrat-  
66 ing particle (vibrating mass element) in the  $i$ th layer of the  
67 body,  $a_{i-1} \leq r \leq a_i$ . Let  $\hat{\mathbf{r}}$  be the unit vector in the direction of  
68 increasing  $r$ , so that the position vector of the equilibrium  
69 point  $P(r, \varphi)$  is  $\mathbf{r} = r\hat{\mathbf{r}}$ . Consider the orthogonal unit vector  
70  $\hat{\boldsymbol{\varphi}} = (\partial\mathbf{r}/\partial\varphi)/|\partial\mathbf{r}/\partial\varphi|$ . Let  $v_i\hat{\boldsymbol{\varphi}} + u_i\hat{\mathbf{r}}$  represent the displacement  
71 from the equilibrium position of the vibrating particle in the  
72  $i$ th layer. For simplicity we suppress the subscript  $i$  if no  
73 confusion is expected. The position vector of the vibrating  
74 particle is thus

$$55 \mathbf{R} = (r + u)\hat{\mathbf{r}} + v\hat{\boldsymbol{\varphi}}. \quad (2)$$

75 Now consider an inertial coordinate system  $OXYZ$  with its  
76 origin  $O$  at the center of the disc, where the  $X$ -,  $Y$ -,  $Z$ -axes  
77 initially correspond to the  $x$ -,  $y$ -,  $z$ -axes, respectively.<sup>7</sup> As-  
78 sume that the disk rotates about the  $Z$ -axis with a small con-  
79 stant angular frequency  $\Omega$ . Consequently, the  $z$ -axis and the  
80  $Z$ -axis are identical, but the angle between the  $X$ -axis (which  
81 is fixed in space) and the  $x$ -axis (which is fixed with respect  
82 to the geometry of the disc) increases at a rate  $\Omega$ . The angu-  
83 lar velocity of the disk is thus  
84

$$85 \boldsymbol{\Omega} = \Omega\hat{\mathbf{k}}, \quad (3)$$

86 where  $\hat{\mathbf{k}} = \hat{\mathbf{r}} \times \hat{\boldsymbol{\varphi}}$  is the unit vector in the direction of the posi-  
87 tive  $Z$ -axis. We assume that the angular rate of rotation  $\Omega$  is  
88 substantially smaller than the lowest vibration frequency of  
89 the system. Consequently, we will neglect centrifugal effects  
90 and all other terms of  $O(\Omega^2)$ .

91 An observer in the  $Oxyz$  coordinate system will measure  
92 the unit vectors  $\hat{\mathbf{r}}$  and  $\hat{\boldsymbol{\varphi}}$  to be constants. Hence this observer  
93 will use Eq. (2) to calculate the velocity  $\mathbf{V}^*$  of the vibrating  
94 particle in the rotating framework  $Oxyz$ , by differentiating  $\mathbf{R}$ ,  
95 treating  $\hat{\mathbf{r}}$  and  $\hat{\boldsymbol{\varphi}}$  as constants:

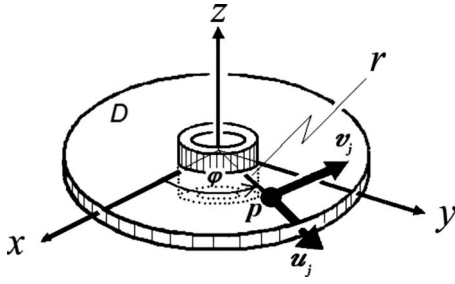


Fig. 1. Coordinate system for the annular disk consisting of various concentric annular layers of varying thickness.

$$\mathbf{V}^* = \frac{d\mathbf{R}}{dt} \Big|_{\hat{\mathbf{r}}, \hat{\boldsymbol{\phi}} = \text{const}} = \dot{u}\hat{\mathbf{r}} + \dot{v}\hat{\boldsymbol{\phi}}. \quad (4)$$

An observer in the  $OXYZ$  coordinate system will note that the direction of the unit vectors  $\hat{\mathbf{r}}$  and  $\hat{\boldsymbol{\phi}}$  is continuously changing. Hence, they are not constant vectors in the  $OXYZ$  frame.

In addition to the velocity  $\mathbf{V}^*$ , we must take into account the velocity imparted by rotation. Recall that a particle moving along a circular path of radius  $r$  and with angular rotation rate  $\omega$  has a tangential speed  $\omega r$ . Hence, a particle with angular velocity  $\boldsymbol{\Omega}$  and position vector  $\mathbf{R}$  has a velocity component given by the cross product  $\boldsymbol{\Omega} \times \mathbf{R}$ . Consequently, the “true velocity” of the vibrating particle as observed from within the fixed frame  $OXYZ$  is

$$\mathbf{V} = \mathbf{V}^* + \boldsymbol{\Omega} \times \mathbf{R} \quad (5)$$

$$= (\dot{u} - \Omega v)\hat{\mathbf{r}} + [\dot{v} + \Omega(r + u)]\hat{\boldsymbol{\phi}}. \quad (6)$$

Spiegel<sup>7</sup> provides a detailed discussion of the derivation of Eq. (5).

### III. KINETIC AND POTENTIAL ENERGY

If we use Eq. (6), the kinetic energy  $E_k$  of the system of particles forming the concentric annular layers is given by

$$E_k = \frac{1}{2} \sum_{i=1}^N \rho_i h_i \int_0^{2\pi} \int_{a_{i-1}}^{a_i} \mathbf{V}_i \cdot \mathbf{V}_i r dr d\varphi \quad (7a)$$

$$\approx \frac{1}{2} \sum_{i=1}^N \rho_i h_i \int_0^{2\pi} \int_{a_{i-1}}^{a_i} [(\dot{u}_i^2 + \dot{v}_i^2) + 2\Omega(u_i \dot{v}_i - \dot{u}_i v_i) + 2\Omega \dot{v}_i r] r dr d\varphi. \quad (7b)$$

When a spring is stretched, the elastic forces involved can do work. Elastic forces are present when an “elastic” body vibrates, and so it is necessary to introduce some of the theory of elasticity to calculate the potential energy of the system of particles forming the concentric annular layers. A short discussion of elasticity is given in Appendixes A–C. According to Eq. (B4), the potential energy  $E_p$  of a system of concentric annular layers is given by

$$E_p = \frac{1}{2} \sum_{i=1}^N h_i \int_0^{2\pi} \int_{a_{i-1}}^{a_i} [\sigma_{r,i} \epsilon_{r,i} + \sigma_{\varphi,i} \epsilon_{\varphi,i} + \tau_{r\varphi,i} \gamma_{r\varphi,i}] r dr d\varphi, \quad (8)$$

where the symbols  $\sigma$  and  $\tau$  stand for the tensile stress and

shear stress, respectively, and  $\epsilon$  and  $\gamma$  stand for tensile strain, and shear strain respectively. According to Eqs. (C3) and (C4), the stresses are

$$\sigma_{r,i} = \frac{E_i}{1 - \nu_i^2} (\epsilon_{r,i} + \nu_i \epsilon_{\varphi,i}), \quad \sigma_{\varphi,i} = \frac{E_i}{1 - \nu_i^2} (\epsilon_{\varphi,i} + \nu_i \epsilon_{r,i}), \quad (9)$$

$$\tau_{r\varphi,i} = \frac{E_i}{2(1 + \nu_i)} \gamma_{r\varphi,i}. \quad (10)$$

Strains may be calculated as follows (see, for instance, Ref. 10 or 11):

$$\epsilon_{r,i} = \frac{\partial u_i}{\partial r}, \quad \epsilon_{\varphi,i} = \frac{1}{r} \left( \frac{\partial v_i}{\partial \varphi} + u_i \right), \quad (11)$$

$$\gamma_{r\varphi,i} = \frac{\partial v_i}{\partial r} + \frac{1}{r} \left( \frac{\partial u_i}{\partial \varphi} - v_i \right). \quad (12)$$

**Problem 1.** Substitute Eqs. (9) and (10) into Eq. (8), and then use Eqs. (11) and (12) to obtain

$$E_p = \frac{1}{2} \sum_{i=1}^N \frac{E_i h_i}{1 - \nu_i^2} \int_0^{2\pi} \int_{a_{i-1}}^{a_i} \left\{ \left[ \frac{\partial u_i}{\partial r} \right]^2 + \left[ \frac{1}{r} \left( \frac{\partial v_i}{\partial \varphi} + u_i \right) \right]^2 + \frac{2\nu_i}{r} \frac{\partial u_i}{\partial r} \left( \frac{\partial v_i}{\partial \varphi} + u_i \right) + \frac{1 - \nu_i}{2} \left[ \frac{\partial v_i}{\partial r} + \frac{1}{r} \left( \frac{\partial u_i}{\partial \varphi} - v_i \right) \right]^2 \right\} r dr d\varphi. \quad (13)$$

### IV. GYROSCOPIC EFFECTS IN DISTRIBUTED BODIES

Equations of motion for the vibrating particle in the  $i$ th body can be obtained by using Eqs. (9)–(12), and the equations of motion discussed by Redwood.<sup>8</sup> The resulting equations consist of two coupled partial differential equations involving terms such as  $\partial^2 u_i / \partial t^2$ ,  $(\partial^2 v_i / \partial t^2)$ ,  $(\partial u_i / \partial r)$ ,  $\partial v_i / \partial \varphi$ ,  $\partial^2 u_i / \partial r \partial \varphi$ . Solving this coupled system of partial differential equations is a nontrivial problem that involves finding, for each  $i$ , two families of eigenfunctions  $U_{i,m}(r)$  and  $V_{i,m}(r)$ ,  $m=2,3,4,\dots$ . The number  $m$  is the vibration mode number or the circumferential wave number. We will not attempt to determine these eigenfunctions here, and we leave this determination for a future paper. We will assume that we can calculate these eigenfunctions and that we can (for each mode of vibration) express the displacements  $u_i$  and  $v_i$  of a vibrating particle in the  $i$ th layer of the body as follows:

$$u_i(r, \varphi, t) = U_i(r)[C(t)\cos m\varphi + S(t)\sin m\varphi], \quad (14)$$

$$v_i(r, \varphi, t) = V_i(r)[C(t)\sin m\varphi - S(t)\cos m\varphi], \quad (15)$$

where the functions  $C(t)$  and  $S(t)$  are to be determined. Here, for simplicity, we have suppressed the mode number on the eigenfunctions, that is,  $U_i(r) = U_{i,m}(r)$  and  $V_i(r) = V_{i,m}(r)$ . It is left as an exercise to determine the nature of the functions  $C(t)$  and  $S(t)$ .

**Problem 2.** Substitute Eqs. (14) and (15) into Eqs. (7b) and (13). Simplification of these expressions involves a long

170 algebraic calculation. The use of a computer algebra system  
171 such as Mathematica or Maple yields

$$172 \quad E_k = \pi[I_0(\dot{C}^2 + \dot{S}^2) + 2\Omega I_1(\dot{C}S - C\dot{S})] \quad (16)$$

173 and

$$174 \quad E_p = \pi I_2(C^2 + S^2), \quad (17)$$

175 where

$$176 \quad I_0 = \frac{1}{2} \sum_{i=1}^N h_i \rho_i \int_{a_{i-1}}^{a_i} (U_i^2 + V_i^2) r \, dr, \quad (18)$$

$$177 \quad I_1 = \sum_{i=1}^N h_i \rho_i \int_{a_{i-1}}^{a_i} U_i V_i r \, dr, \quad (19)$$

178 and

$$179 \quad I_2 = \frac{1}{2} \sum_{i=1}^N \frac{E_i h_i}{1 - \nu_i^2} \int_{a_{i-1}}^{a_i} \left\{ (U_i')^2 + 2\nu_i U_i' \frac{U_i + mV_i}{r} \right. \\ \left. + \left( \frac{U_i + mV_i}{r} \right)^2 + \frac{1 - \nu_i}{2} \left( V_i' - \frac{mU_i + V_i}{r} \right)^2 \right\} r \, dr. \quad (20)$$

180

### 181 A. Lagrange's equations

182 The Lagrangian follows from Eqs. (16) and (17):

$$183 \quad L(C, S, \dot{C}, \dot{S}) = E_k - E_p = \pi[I_0(\dot{C}^2 + \dot{S}^2) - 2\Omega I_1(C\dot{S} - \dot{C}S) \\ 184 \quad - (C^2 + S^2)I_2]. \quad (21)$$

185 The vibration of the  $m$ th mode is governed by Lagrange's  
186 equations of motion:

$$187 \quad \frac{d}{dt} \left( \frac{\partial L}{\partial \dot{C}} \right) - \frac{\partial L}{\partial C} = 0, \quad (22)$$

188 and

$$189 \quad \frac{d}{dt} \left( \frac{\partial L}{\partial \dot{S}} \right) - \frac{\partial L}{\partial S} = 0. \quad (23)$$

190 Equations (22) and (23) yield

$$191 \quad \ddot{C} + 2\eta\Omega\dot{S} + \omega^2 C = 0 \quad (24)$$

192 and

$$193 \quad \ddot{S} - 2\eta\Omega\dot{C} + \omega^2 S = 0, \quad (25)$$

194 respectively, where, for the  $m$ th mode of vibration

$$195 \quad -1 \leq \eta = \frac{I_1}{I_0} \leq 1, \quad (26)$$

196 and  $\omega$  is given by

$$197 \quad \omega = \sqrt{\frac{I_2}{I_0}}. \quad (27)$$

### 198 B. Bryan's factor

199 We now show that  $\omega$  is an eigenvalue of the vibrating  
200 system and that  $\eta$  in Eq. (26) is Bryan's factor BF in Eq. (1).

To interpret what Eqs. (24) and (25) represent we combine  
the two equations by considering the complex function

$$Z = C + iS \quad (28) \quad 203$$

to obtain the single equation

$$\ddot{Z} - i(2\eta\Omega)\dot{Z} + \omega^2 Z = 0. \quad (29) \quad 205$$

If we write  $Z$  in polar form

$$Z(t) = r(t)e^{i\theta(t)}, \quad (30) \quad 207$$

and assume that  $\theta(t)$  has the linear form  $\theta(t) = at$ , we obtain

$$\ddot{r} + 2i(a - \eta\Omega)\dot{r} + (2\eta\Omega a - a^2 + \omega^2)r = 0. \quad (31) \quad 209$$

If we choose  $a = \eta\Omega$ , the coefficient of  $\dot{r}$  is eliminated in Eq. (31), and we obtain the differential equation of a harmonic oscillator:

$$\ddot{r} + \gamma^2 r = 0, \quad (32) \quad 213$$

where

$$\gamma = \sqrt{\omega^2 + \eta^2 \Omega^2} \quad (33) \quad 215$$

is an eigenvalue of the vibrating system with eigenfrequency  
of vibration  $f = \gamma/2\pi$ . According to the assumption made  
shortly after Eq. (3),  $\Omega \ll f$ . Consequently,

$$\gamma \approx \omega, \quad (34) \quad 219$$

and so  $\omega$  is an eigenvalue of the vibrating system. Equations  
(24) and (25) can now be viewed in the form

$$Z(t) = r(t)e^{i\eta\Omega t}. \quad (35) \quad 222$$

Equation (35) shows that Eqs. (24) and (25) represent a "vec-  
tor" in the complex plane with its magnitude varying like a  
harmonic oscillator and its position varying at a rate propor-  
tional to the constant, small rotation rate  $\Omega$  of the isotropic  
body. Hence, according to Eq. (1), Bryan's factor

$$\text{BF} = \frac{\eta\Omega}{\omega} = \eta. \quad (36) \quad 228$$

Consequently, if a gyroscope based on Bryan's effect<sup>12</sup> is to  
be calibrated, then, without conducting lengthy experiments,  
Bryan's factor can be calculated from Eq. (26) once the  
eigenfunctions of Eqs. (14) and (15) are known.

Equations (18)–(20), (26), and (27) show that for the  $m$ th  
mode of vibration, Bryan's factor and the eigenfrequency of  
vibration depend on physical properties such as the density  
and geometrical properties such as thickness. The eigenfre-  
quency also depends on elastic properties such as Young's  
modulus and Poisson's ratio.

Equation (35) defines a precessing wave. The rotating vi-  
bration pattern lags behind the position of the static vibra-  
tion pattern if  $\eta < 0$  and precedes the position of the static vibra-  
tion pattern if  $\eta > 0$ . A calculation of  $\eta$  for a liquid filled  
wineglass<sup>3</sup> and  $m=2$  reveals  $\eta$  to be negative. Hence, the  
rotating vibration pattern should lag behind the static vibra-  
tion pattern for the wineglass.

We note that Eqs. (24) and (25) are obtained with appro-  
priate values of  $I_0$ ,  $I_1$ , and  $I_2$  for isotropic cylindrical or  
spherical distributed bodies. The definite integrals  $I_0$ ,  $I_1$ , and  
 $I_2$  are far more complicated for a cylinder and sphere.

250 **Problem 3.** Show that to a good approximation

251  $C(t) = \cos \eta\Omega t(A \cos \omega t + B \sin \omega t),$  (37)

252  $S(t) = \sin \eta\Omega t(A \cos \omega t + B \sin \omega t)$  (38)

253 (where  $A$  and  $B$  are arbitrary constants) by solving Eq. (32)  
254 for  $r(t)$ , substituting into Eq. (35), equating real and imagi-  
255 nary parts, and then using Eq. (34).

256 **Problem 4.** Use the Lagrangian  $L$  as given by Eq. (21) and  
257 include viscous damping by introducing Rayleigh's dissipa-  
258 tion function  $\mathcal{F}=(c\dot{C}^2+s\dot{S}^2)/2$  into Lagrange's equations  
259 (see Ref. 9). Assume weak, isotropic, viscous damping, that  
260 is,  $c=s=\pi D$ , with the damping factor  $\delta=D/(2I_0)$  much  
261 smaller than the lowest eigenfrequency of the vibrating sys-  
262 tem. Conclude that the introduction of light, viscous, isotro-  
263 pic damping into the considerations does not alter the fact  
264 that the damped vibrating pattern rotates at a rate  $\eta\Omega$  in the  
265  $Oxyz$  plane, where  $\eta$  is given by Eq. (26). See Ref. 6 for  
266 details.

## 267 V. CONCLUSION

268 By using standard concepts of physics such as kinetic en-  
269 ergy, potential energy, and Lagrange's equations, we have  
270 demonstrated how Bryan's effect for a composite disk that is  
271 rotating slowly in space can be predicted and Bryan's factor  
272 can be calculated. These considerations also demonstrated  
273 that Bryan's factor depends on properties such as the density  
274 and the thickness of the disk and that the eigenfrequency of  
275 vibration of the disk also depends on elastic properties such  
276 as Young's modulus and Poisson's ratio.

277 We can now better understand the operation and calibra-  
278 tion of the hemispherical resonator gyroscope of Loper and  
279 Lynch.<sup>12</sup> Roughly speaking, suppose that a vibrating hemi-  
280 sphere is fixed to a vehicle (such as a space shuttle or sub-  
281 marine) moving through three-dimensional space and that a  
282 sensor inside the vehicle observes the position of a node of  
283 the fundamental vibration of the hemisphere (such vibrations  
284 can be observed in the excellent holographic interferograms  
285 of a vibrating wineglass in Ref. 13). Suppose the vehicle  
286 undergoes a slow rate of rotation  $\Omega$  with respect to the space  
287 through which it is moving and that this rotation rate is too  
288 small for the human vestibular system to observe. The sensor  
289 will register that the node rotates away from its original po-  
290 sition. From observations within the vehicle the rotation rate  
291  $\alpha$  of the node can be calculated and, using Bryan's factor  $\eta$   
292 for the fundamental mode of vibration, the rate of rotation of  
293 the vehicle  $\Omega=\alpha/\eta$  with respect to the space through which  
294 it is moving can be calculated.

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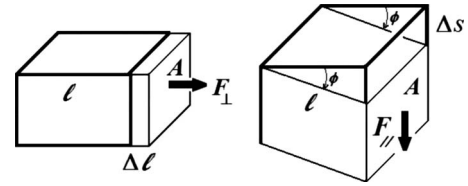


Fig. 2. Elastic blocks undergoing tensile deformation (left block) and shear deformation (right block).

## APPENDIX A: ELASTIC CONSTANTS

300

A body will deform when stretching and/or twisting forces  
are applied to it. We consider as a first approximation a per-  
fectly elastic body that returns to its original form after  
stretching and/or twisting forces are removed from it.

Consider a length  $\ell$  of an elastic block (Fig. 2) with cross-  
sectional area  $A$  that is subjected to a stretching force  $F_{\perp}$   
(normal to the area  $A$ ) causing the side length to increase  
from  $\ell$  to  $\ell + \Delta\ell$ . The tensile stress  $\sigma$  of the elastic body is  
given by

$$\sigma = \frac{F_{\perp}}{A}, \quad (\text{A1}) \quad 310$$

and the tensile strain  $\epsilon$  is given by

$$\epsilon = \frac{\Delta\ell}{\ell}. \quad (\text{A2}) \quad 312$$

Young's modulus (or the modulus of elasticity)  $E$  is given by

$$E = \frac{\text{tensile stress}}{\text{tensile strain}} = \frac{\sigma}{\epsilon}. \quad (\text{A3}) \quad 314$$

If a length of elastic body is stretched from length  $\ell$  to  $\ell + \Delta\ell$ , its transverse dimensions (its height or breadth)  $t$  decreases from  $t$  to  $t + \Delta t$  (here  $\Delta t < 0$ ). Hence, both longitudinal strain  $\Delta\ell/\ell$  and transverse strain  $\Delta t/t$  are present simultaneously. (For isotropic substances transverse strain is the same for any transverse dimensions such as height, breadth, or diameter.) Poisson's ratio  $\nu$  is defined as the positive dimensionless constant

$$\nu = - \left( \frac{\text{longitudinal strain}}{\text{transverse strain}} \right) = - \left( \frac{\Delta\ell/\ell}{\Delta t/t} \right). \quad (\text{A4}) \quad 323$$

Suppose that an elastic block (Fig. 2) is subjected to a  
shearing or twisting force  $F_{\parallel}$  (parallel to the area  $A$ ) that  
twists the body through a small angle  $\phi$  (in Fig. 2,  $\phi \approx \Delta s/\ell$ ). The shear stress  $\tau$  of the body is given by

$$\tau = \frac{F_{\parallel}}{A}, \quad (\text{A5}) \quad 328$$

and the shear strain  $\gamma$  is given by

$$\gamma = \phi. \quad (\text{A6}) \quad 330$$

The shear modulus  $G$  is given by

$$G = \frac{\text{shear stress}}{\text{shear strain}} = \frac{\tau}{\gamma}. \quad (\text{A7}) \quad 332$$

It can be shown (see Ref. 10) that

333



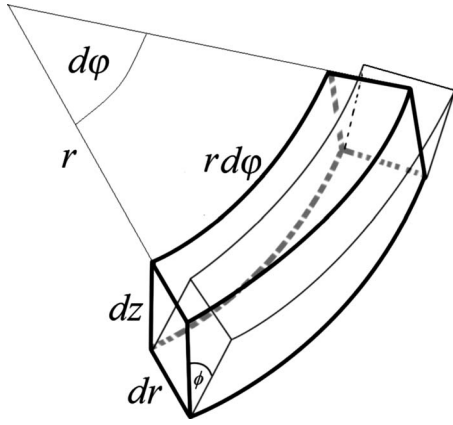


Fig. 3. Volume element  $dV=rdrd\phi dz$  in polar coordinates before deformation (thick lines) and after deformation (thin lines).

there is a shear stress  $\tau_{r\phi}$  parallel to the  $r\phi$ -plane. Suppose that the volume element  $dV$  is subjected to a shear force  $\tau_{r\phi}rd\phi dr$  which produces a shear extension  $\gamma_{r\phi}dz$ . According to the shear version of Eq. (B1b), the work done by the shear force to produce this shear extension is

$$\frac{1}{2}(\text{shear force})(\text{shear extension}) = \frac{1}{2}(\tau_{r\phi}rd\phi dr)[\gamma_{r\phi}dz]. \quad (\text{B3})$$

There are two similar expressions for the work done by the radial and tangential tensile forces. We sum the three expressions to obtain the total potential (or strain) energy of the volume element:

$$dW = \frac{1}{2}(\sigma_r\epsilon_r + \sigma_\phi\epsilon_\phi + \tau_{r\phi}\gamma_{r\phi})rdrd\phi dz. \quad (\text{B4})$$

### APPENDIX C: SUPERPOSITION

Consider the tensile and shear strains of the volume element  $dV$  of Eq. (B2). A radial (tangential) strain  $\sigma_r/E$  ( $\sigma_\phi/E$ ) is accompanied by a lateral contraction per unit length or a lateral strain in the tangential (radial) direction  $-\nu\sigma_r/E$  ( $-\nu\sigma_\phi/E$ ), where  $\nu$  is Poisson's ratio. Shear stresses do not cause lateral stresses. Hence, by superposition, the net strains are

$$\epsilon_r = \frac{1}{E}(\sigma_r - \nu\sigma_\phi), \quad \epsilon_\phi = \frac{1}{E}(\sigma_\phi - \nu\sigma_r), \quad (\text{C1})$$

$$\gamma_{r\phi} = \frac{1}{G}\tau_{r\phi} = \frac{2(1+\nu)}{E}\tau_{r\phi}. \quad (\text{C2})$$

**Problem 5.** Solve Eq. (C1) simultaneously and manipulate Eq. (C2) to show that stresses are given in terms of strains as

$$\sigma_r = \frac{E}{1-\nu^2}(\epsilon_r + \nu\epsilon_\phi), \quad \sigma_\phi = \frac{E}{1-\nu^2}(\epsilon_\phi + \nu\epsilon_r), \quad (\text{C3})$$

$$\tau_{r\phi} = \frac{E}{2(1+\nu)}\gamma_{r\phi}. \quad (\text{C4})$$

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$$G = \frac{E}{2(1+\nu)}, \quad (\text{A8})$$

and so

$$\gamma = \frac{2(1+\nu)}{E}\tau. \quad (\text{A9})$$

Equations (A3) and (A7) are forms of Hooke's law. For instance, from Eq. (A3), we deduce the well known law for elastic elongation  $w$  (usually referred to as Hooke's law):

$$F = kw, \quad (\text{A10})$$

with  $k=AE/\ell$ ,  $w=\Delta\ell$ , and  $F=F_\perp$ .

### APPENDIX B: POTENTIAL ENERGY

When a tensile extension  $w$  occurs, work has been done by the force  $F=F_\perp$ . This work is stored as potential energy. According to the tensile form of Hooke's law given by Eq. (A10), this tensile potential energy (also called tensile strain energy) is given by

$$\int_0^w F dw = k \int_0^w w dw = \frac{1}{2}(kw)[w] = \frac{1}{2}(F_\perp) \times [\Delta\ell] \quad (\text{B1a})$$

$$= \frac{1}{2}(\text{tensile force}) \times [\text{tensile extension}]. \quad (\text{B1b})$$

A similar formula holds for shear potential energy. In an elastic solid disk with distributed parameters as depicted in Fig. 1, suppose that we have an elastic volume element  $dV$  at the point  $P$ ,

$$dV = drdzrd\phi = rdrd\phi dz, \quad (\text{B2})$$

as depicted in Fig. 3. Here the thickness of the disk at point  $P$  is  $h = \int_0^h dz$ .

Tensile stresses  $\sigma_r$  in the radial direction and  $\sigma_\phi$  in the tangential direction exist, but there are no tensile stresses on faces (areas) parallel to the area  $rd\phi dr$  in the  $r\phi$ -plane. There are no shear stresses parallel to the areas  $drdz$  or  $rd\phi dz$ , but

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